

(Almost) Global UAV Control System Design

Aided by Lyapunov, Barbalat, and Matrosov

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5/21-2021

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Introduction

Difficulties

- Nonlinear
- Nonautonomous
- Configuration manifolds

Main idea

- Example 1 - Peaking
- Example 2 - Uniform stability
- Example 3 - Case study



Figure 1: The small Crazyflie quadrotor

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Figure 1: The small Crazyflie quadrotor

Pontus: Good idea to talk about Lyapunov functions!

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Figure 1: The small Crazyflie quadrotor

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[1] E. Lefeber, M. Greiff, and A. Robertsson, "Filtered output feedback tracking control of a quadrotor UAV," *IFAC-PapersOnLine*, vol. 53, no. 2, pp. 5764–5770, 2020

Overview and main thread of presentation

- Introduction
 - Pitfalls in NLTV analysis
 - Introducing the case study
- Lyapunov's Second Method
 - The main idea
- Barbălat's Lemma
 - The main idea
 - Useful variations
 - Application
- Matrosov's Theorems
 - The main idea
 - Application
- Simulation example

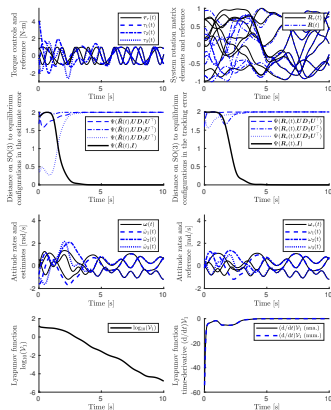


Figure 2: Example simulation (to be explained)

Introduction - A note of caution

Some warnings

- Quite dense
- Lots of signals...!
- Some omitted details

Σ A system (with memory)

R A rotation (always $\in \mathbb{R}^{3 \times 3}$)

r A reference

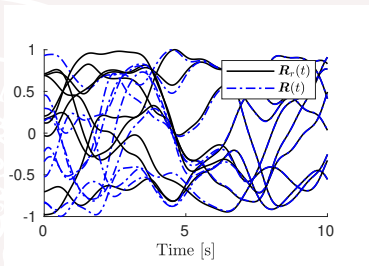
e A tracking error

G Global, as in *globally* stable (GS)

A Asymptotic, as in *asymptotically* stable (AS)

E Exponential, as in *exponentially* stable (ES)

U Uniform, as in *uniformly* stable (US)

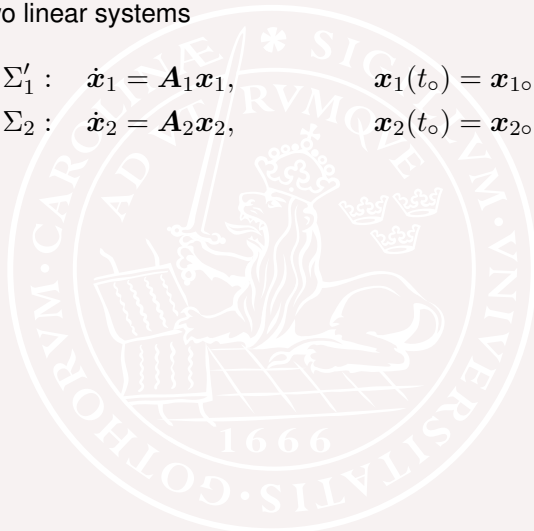


Introduction - Curious Example 1

Consider two linear systems

$$\Sigma'_1 : \quad \dot{\mathbf{x}}_1 = \mathbf{A}_1 \mathbf{x}_1, \quad \mathbf{x}_1(t_0) = \mathbf{x}_{10} \quad (1a)$$

$$\Sigma_2 : \quad \dot{\mathbf{x}}_2 = \mathbf{A}_2 \mathbf{x}_2, \quad \mathbf{x}_2(t_0) = \mathbf{x}_{20}. \quad (1b)$$



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$$\Sigma_1 : \quad \dot{\mathbf{x}}_1 \triangleq \mathbf{A}_1 \mathbf{x}_1 + \mathbf{B} \mathbf{x}_2. \quad (2)$$

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If $\{\Sigma'_1, \Sigma_2\}$ are asymptotically stable (AS), then $\{\Sigma_1, \Sigma_2\}$ is AS, as

$$\{\Sigma_1, \Sigma_2\} : \quad \frac{d}{dt} \begin{bmatrix} \mathbf{x}_1 \\ \mathbf{x}_2 \end{bmatrix} = \begin{bmatrix} \mathbf{A}_1 & \mathbf{B} \\ \mathbf{0} & \mathbf{A}_2 \end{bmatrix} \begin{bmatrix} \mathbf{x}_1 \\ \mathbf{x}_2 \end{bmatrix} \quad (3)$$

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What if the systems are nonlinear and non-autonomous?

Introduction - Curious Example 1

Consider two nonlinear systems

$$\Sigma'_1 : \quad \dot{\mathbf{x}}_1 = f_1(t, \mathbf{x}_1), \quad \mathbf{x}_1(t_0) = \mathbf{x}_{10} \quad (4a)$$

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$$\Sigma_1 : \quad \dot{\mathbf{x}}_1 \triangleq f_1(t, \mathbf{x}_1) + g(t, \mathbf{x}_1, \mathbf{x}_2)\mathbf{x}_2. \quad (5)$$

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If $\{\Sigma'_1, \Sigma_2\}$ are asymptotically stable (AS), then $\{\Sigma_1, \Sigma_2\}$ is...

...it depends!

Introduction - Curious Example 1

Example (Peaking)

Let $t_o = 0$, and consider a nonlinear system

$$\Sigma'_1 : \dot{x}_1 = -x_1^3,$$

$$x_1(t_o) = x_{1o},$$

$$\Sigma_2 : \dot{x}_2 = -x_2,$$

$$x_2(t_o) = x_{2o}.$$

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The solution for the system $\{\Sigma_1, \Sigma_2\}$ is

$$\begin{aligned}x_1(t) &= \text{sign}(x_{1o})(x_{1o}^{-2} + 2x_{2o}(e^{-t} - 1) + 2t)^{-1/2} \\ x_2(t) &= x_{2o}e^{-t},\end{aligned}$$

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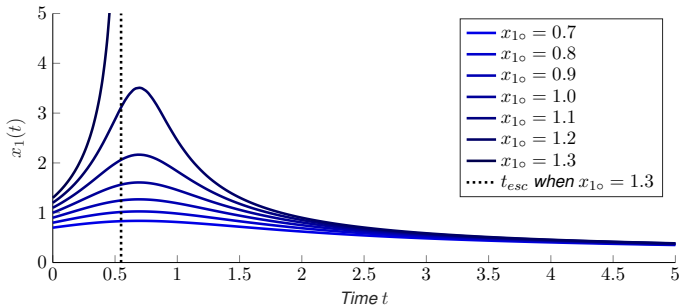
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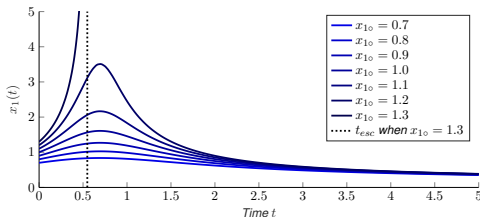
Here Σ'_1 is GAS, and Σ_2 is GAS, but for their cascade through g , the solution $x_1(t)$ diverges with a finite escape time even for $x_{1o} > 0$.



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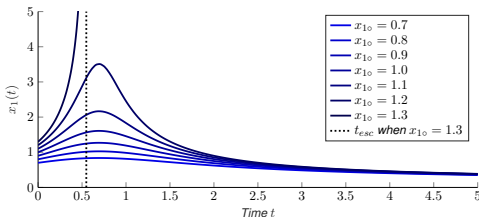


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Introduction - Curious Example 1

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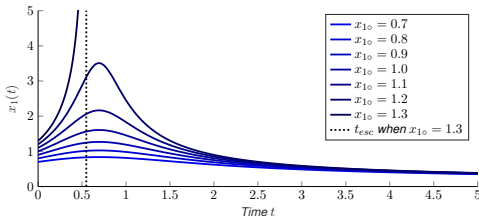
If $\{\Sigma'_1, \Sigma_2\}$ are asymptotically stable (AS), then $\{\Sigma_1, \Sigma_2\}$ is...

GAS if a set of sufficient conditions on $\{\Sigma'_1, \Sigma_2\}$ and g are met.

Introduction - Curious Example 1

Example (Peaking)

Here Σ_1' is GAS, and Σ_2 is GAS, but for their cascade through g , the solution $x_1(t)$ diverges with a finite escape time even for $x_{10} > 0$.



Growth rate: If $\{\Sigma_1', \Sigma_2\}$ is AS, and there exists continuous $\theta_i : \mathbb{R}_{\geq 0} \rightarrow \mathbb{R}_{\geq 0}$ such that $\|g(x_1, x_2)\| \leq \theta_1(\|x_2\|) + \theta_2(\|x_2\|)\|x_1\|$, then $\{\Sigma_1, \Sigma_2\}$ is also AS (see e.g. Panteley '99 [2], or Loria '05 [3]).

Introduction - Curious Example 1

Takeaway

In the nonlinear setting, the separation principle generally does not apply. Care must be taken when connecting a found controller with an observer, as the introduced dynamics may cause the states to diverge, even if $\{\Sigma'_1, \Sigma_2\}$ has very nice properties. Especially true when aiming for global or almost global stability properties.

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In the nonlinear setting, the separation principle generally does not apply. Care must be taken when connecting a found controller with an observer, as the introduced dynamics may cause the states to diverge, even if $\{\Sigma'_1, \Sigma_2\}$ has very nice properties. Especially true when aiming for global or almost global stability properties.

With a "nice" feedback, a "nice" estimator and "good" connection, are asymptotic stability properties enough? What about robustness?

Introduction - Curious Example 2

Consider two *non-autonomous* systems:

$$\Sigma : \quad \dot{\mathbf{x}} = f(t, \mathbf{x}),$$

$$\Sigma_{\Delta} : \quad \dot{\mathbf{x}} = f(t, \mathbf{x}) + \mathbf{\Delta}(t, \mathbf{x}),$$

where $\|\mathbf{\Delta}(t, \mathbf{x})\| \leq L$ for all $t \geq t_0$. What does Σ say about Σ_{Δ} ?

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Consider two *non-autonomous* systems:

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Example (Loria, Panteley, Teel '99)

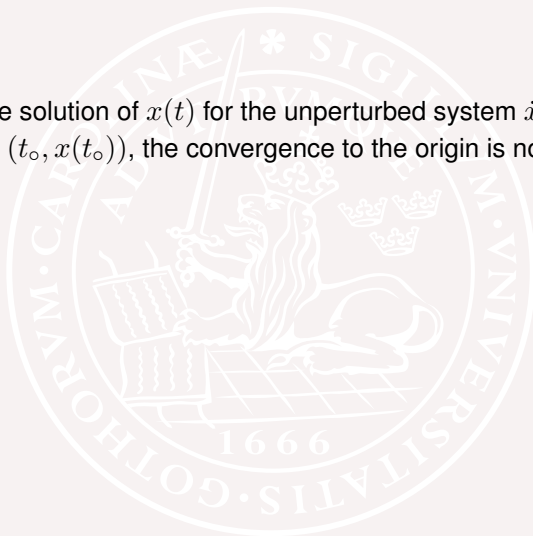
Consider a nominal system with defined by $a(t) = (t + 1)^{-1}$, with

$$\dot{x} = f(t, x) = \begin{cases} -a(t)\text{sign}(x) & \text{if } |x| > a(t) \\ -x & \text{if } |x| \leq a(t) \end{cases}, \quad t \geq t_0 \geq 0.$$

When adding a $\Delta(t) = L \neq 0$, solutions grow unbounded as $t \rightarrow \infty$.
In fact, $\lim_{t \rightarrow \infty} x(t)/t = \pm L$ (depending on the sign of $x(t_0)$)

Introduction - Curious Example 2

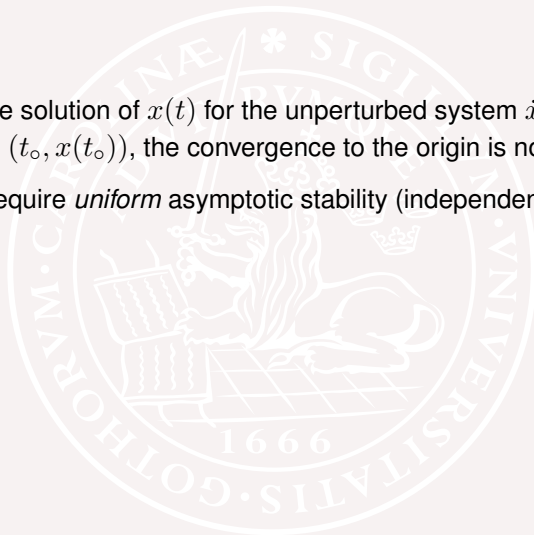
Problem: the solution of $x(t)$ for the unperturbed system $\dot{x} = f(t, x)$ depends on $(t_o, x(t_o))$, the convergence to the origin is not *uniform*.



Introduction - Curious Example 2

Problem: the solution of $x(t)$ for the unperturbed system $\dot{x} = f(t, x)$ depends on $(t_o, x(t_o))$, the convergence to the origin is not *uniform*.

Solution: Require *uniform* asymptotic stability (independent of t_o).



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Takeaway

In general, to say something about the robustness properties of systems on the form $\Sigma : \dot{x} = f(t, x)$ we require uniform stability properties. Several local or global boundedness results follow (see e.g., Khalil '96 [4, Theorem 3.18 combined with Lemma 4.3]).

Introduction

The attitude dynamics of the UAV

$$\Sigma : \begin{cases} \dot{\mathbf{R}} &= \mathbf{R}\mathbf{S}(\boldsymbol{\omega}), \\ \mathbf{J}\dot{\boldsymbol{\omega}} &= \mathbf{S}(\mathbf{J}\boldsymbol{\omega})\boldsymbol{\omega} + \boldsymbol{\tau}, \end{cases} \quad \text{(Controlled system)} \quad (6a)$$

$$\Sigma_r : \begin{cases} \dot{\mathbf{R}}_r &= \mathbf{R}_r\mathbf{S}(\boldsymbol{\omega}_r), \\ \mathbf{J}\dot{\boldsymbol{\omega}}_r &= \mathbf{S}(\mathbf{J}\boldsymbol{\omega}_r)\boldsymbol{\omega}_r + \boldsymbol{\tau}_r, \end{cases} \quad \text{(Reference system)} \quad (6b)$$

where

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where

$$\begin{aligned} \mathbf{J} &\in \mathbb{R}^{3 \times 3} && \text{s.t.} && \mathbf{J} = \mathbf{J}^\top \succ \mathbf{0} \\ \mathbf{S} : \mathbb{R}^3 &\rightarrow \mathbb{R}^{3 \times 3} && \text{s.t.} && \mathbf{S}(\mathbf{a})\mathbf{b} = \mathbf{a} \times \mathbf{b} \end{aligned}$$

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Introduction

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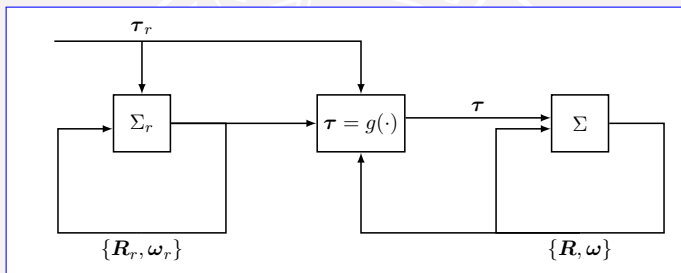
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$$\boldsymbol{\tau}, \boldsymbol{\tau}_r \in \mathbb{R}^3 \quad \text{s.t.} \quad \|\boldsymbol{\tau}_r\| \text{ is uniformly bounded in } t$$

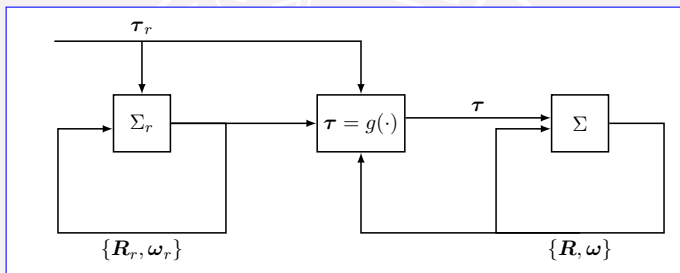
Introduction

Objective: Find $g(\mathbf{R}, \boldsymbol{\omega}, \mathbf{R}_r, \boldsymbol{\omega}_r, \boldsymbol{\tau}_r)$ such that $\mathbf{R} \rightarrow \mathbf{R}_r, \boldsymbol{\omega} \rightarrow \boldsymbol{\omega}_r$



Introduction

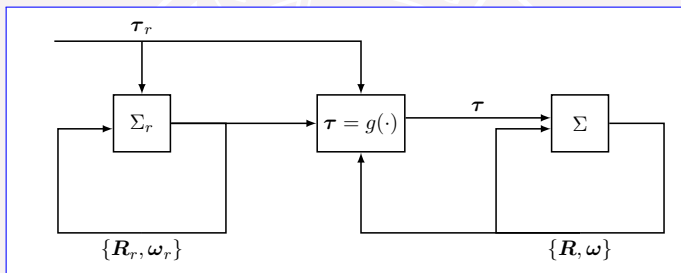
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- Many (almost) global solutions exist [5]–[8], however...

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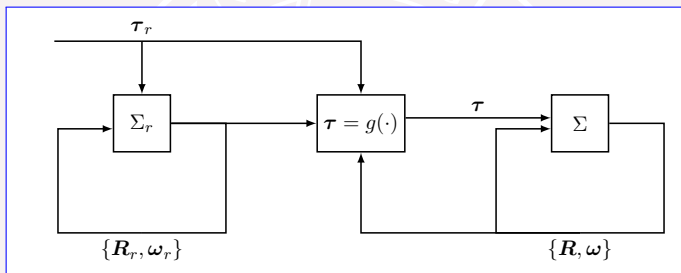
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- Many (almost) global solutions exist [5]–[8], however...
- Essentially a full-state feedback - requires an estimator

Introduction

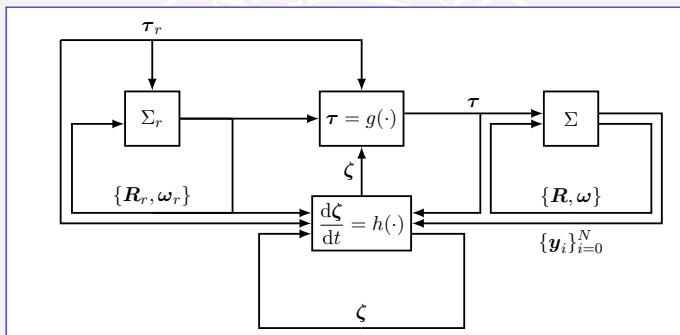
Objective: Find $g(\mathbf{R}, \boldsymbol{\omega}, \mathbf{R}_r, \boldsymbol{\omega}_r, \boldsymbol{\tau}_r)$ such that $\mathbf{R} \rightarrow \mathbf{R}_r, \boldsymbol{\omega} \rightarrow \boldsymbol{\omega}_r$



- Many (almost) global solutions exist [5]–[8], however...
- Essentially a full-state feedback - requires an estimator
- Stability should be uniform, estimator needs to be (almost) globally stabilizing, interconnection needs to satisfy conditions.

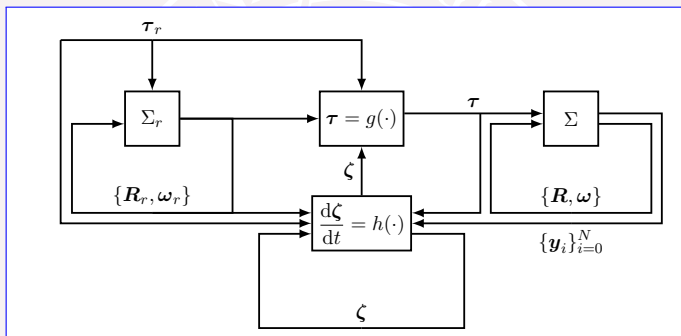
Introduction

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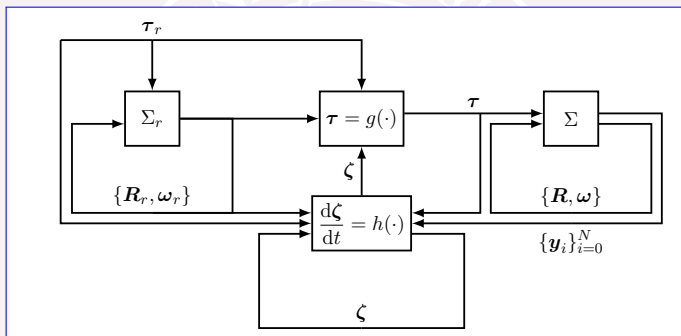
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- Define a filter memory ζ (here $\{\hat{\mathbf{R}}, \hat{\boldsymbol{\omega}}\} \in \text{SO}(3) \times \mathbb{R}^3$).

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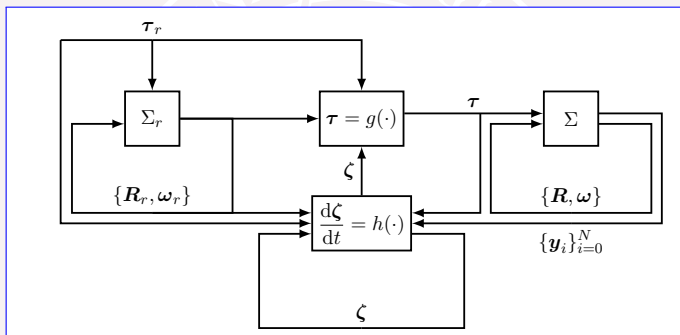
Alternatively, solve a *filtered output feedback problem*, as in [1].



- Define a filter memory ζ (here $\{\hat{\mathbf{R}}, \hat{\boldsymbol{\omega}}\} \in \text{SO}(3) \times \mathbb{R}^3$).
- Define an update of ζ in a set of measurements $\{\mathbf{y}_i\}_{i=1}^N$ (IMU).

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- Define an update of ζ in a set of measurements $\{\mathbf{y}_i\}_{i=1}^N$ (IMU).
- Define a feedback law $g(\zeta, \mathbf{R}_r, \boldsymbol{\omega}_r, \tau_r)$ such that $\mathbf{R} \rightarrow \mathbf{R}_r, \boldsymbol{\omega} \rightarrow \boldsymbol{\omega}_r$ and $\{\zeta, \mathbf{R}, \boldsymbol{\omega}, \tau\}$ remain bounded.

Introduction

Case: *filtered output feedback* in [1].

- Illustrate the ideas
- Less focus on precision
- Defining the errors
- A curious Lyapunov function
- Applying Barbālat
- Applying Matrosov
- Simulation example



Available online at www.elsevier.com/locate/issn/0898-1221
ScienceDirect
ISSN 0898-1221
ISSN 1876-7865



Filtered Output Feedback Tracking Control of a Quadrotor UAV

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Abstract. We present a tracking controller for quadrotor UAVs, which uses spatial error information and filters the measurements to attenuate noise. We show robustness against global asymptotic and local output feedback of the resulting closed-loop system, which requires differentiability against bounded disturbances. We discuss the performance of the controller by means of several numerical examples, including a simple trajectory maneuver.

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Keywords: UAV, tracking, output feedback control, nonlinear, disturbance, Lyapunov methods.

In this paper we consider the problem of asymptotically stabilizing a quadrotor (without a wind) for the tracking control of quadrotor UAVs, without the use of linear velocity measurements. In other words, we use only position and orientation measurements to design the controller, and also for analyzing systems the so-called input-output property does not hold, coordinate observability conditions do not hold, and so on.

Starting with the work of Kawachi and Tsubota (2005), output feedback laws that solve the tracking problem for UAVs have been proposed. In the literature, the tracking problem has been solved by only the position and orientation measurements in a nonadaptive manner. In this paper, we propose a tracking controller that uses only position and orientation measurements to solve the tracking problem. The controller is designed to be robust against bounded disturbances. We show that the resulting closed-loop system is globally asymptotically stable in the sense of Lyapunov. We also show that the controller is robust against bounded disturbances. We discuss the performance of the controller by means of several numerical examples, including a simple trajectory maneuver.

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[1] E. Lefeber, M. Greiff, and A. Robertsson, “Filtered output feedback tracking control of a quadrotor UAV,” *IFAC-PapersOnLine*, vol. 53, no. 2, pp. 5764–5770, 2020

Case study - Errors

Consider the errors,

$$\mathbf{R}_e = \mathbf{R}_r \mathbf{R}^\top \in \text{SO}(3), \quad (7a)$$

$$\tilde{\mathbf{R}} = \hat{\mathbf{R}} \mathbf{R}^\top \in \text{SO}(3), \quad (7b)$$

$$\boldsymbol{\omega}_e = \boldsymbol{\omega}_r - \boldsymbol{\omega} \in \mathbb{R}^3, \quad (7c)$$

$$\tilde{\boldsymbol{\omega}} = \hat{\boldsymbol{\omega}} - \boldsymbol{\omega} \in \mathbb{R}^3, \quad (7d)$$

$$\hat{\boldsymbol{\omega}}_e = \boldsymbol{\omega}_r - \hat{\boldsymbol{\omega}} \in \mathbb{R}^3. \quad (7e)$$

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$$\hat{\boldsymbol{\omega}}_e = \boldsymbol{\omega}_r - \hat{\boldsymbol{\omega}} \in \mathbb{R}^3. \quad (7e)$$

With the controller and observer in [1] (here omitted for brevity),

$$\dot{\mathbf{R}}_e = f_1(t, \mathbf{R}_e, \tilde{\mathbf{R}}_e, \boldsymbol{\omega}_e, \tilde{\boldsymbol{\omega}}) \quad (8a)$$

$$\dot{\tilde{\mathbf{R}}}_e = f_2(t, \mathbf{R}_e, \tilde{\mathbf{R}}_e, \boldsymbol{\omega}_e, \tilde{\boldsymbol{\omega}}) \quad (8b)$$

$$\mathbf{J} \dot{\boldsymbol{\omega}}_e = f_3(t, \mathbf{R}_e, \tilde{\mathbf{R}}_e, \boldsymbol{\omega}_e, \tilde{\boldsymbol{\omega}}) \quad (8c)$$

$$\mathbf{J} \dot{\tilde{\boldsymbol{\omega}}} = f_4(t, \mathbf{R}_e, \tilde{\mathbf{R}}_e, \boldsymbol{\omega}_e, \tilde{\boldsymbol{\omega}}) \quad (8d)$$

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Case Study - Lyapunov

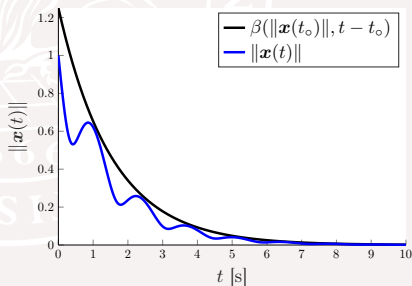
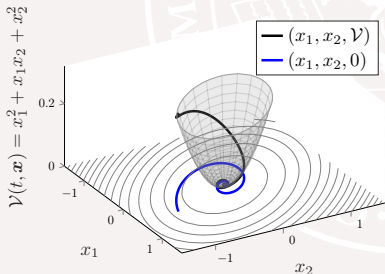
Consider an AS linear system,

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x}, \quad \mathbf{x}(t_0) \in \mathbb{R}^n.$$

Then, there exists a solution

$$\mathbf{A}^\top \mathbf{P} + \mathbf{P}\mathbf{A} + \mathbf{Q} = \mathbf{0}, \quad \mathbf{P} = \mathbf{P}^\top \succ \mathbf{0}, \quad \mathbf{Q} = \mathbf{Q}^\top \succ \mathbf{0},$$

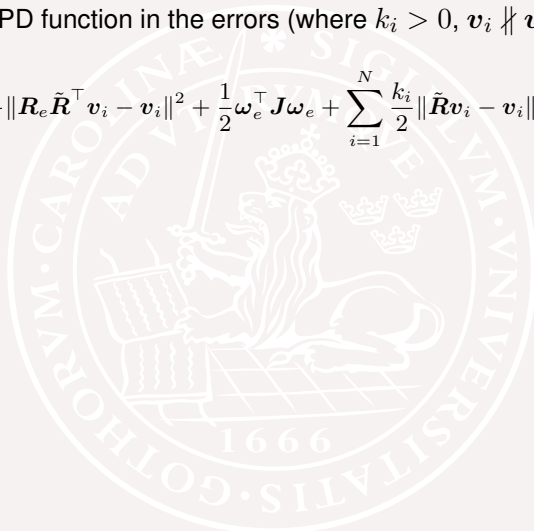
using $\gamma_{\text{Lyp}}(\mathbf{A}, \mathbf{Q})$ and a quadratic Lyapunov function $\mathcal{V} = \mathbf{x}^\top \mathbf{P}\mathbf{x}$.



Case study - Lyapunov

Consider a PD function in the errors (where $k_i > 0$, $\mathbf{v}_i \neq \mathbf{v}_j \in \mathbb{R}^3$),

$$\mathcal{V}_1 = \sum_{i=1}^N \frac{k_i}{2} \|\mathbf{R}_e \tilde{\mathbf{R}}^\top \mathbf{v}_i - \mathbf{v}_i\|^2 + \frac{1}{2} \boldsymbol{\omega}_e^\top \mathbf{J} \boldsymbol{\omega}_e + \sum_{i=1}^N \frac{k_i}{2} \|\tilde{\mathbf{R}} \mathbf{v}_i - \mathbf{v}_i\|^2 + \frac{1}{2} \tilde{\boldsymbol{\omega}}^\top \mathbf{J} \tilde{\boldsymbol{\omega}}.$$



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The function is NSD along the solutions of the error dynamics, with

$$\dot{\mathcal{V}}_1 = -c_R \left\| \sum_{i=1}^N k_i \mathbf{S}(\hat{\mathbf{R}}^\top \mathbf{v}_i) (\mathbf{R}_r^\top \mathbf{v}_i + \mathbf{R}^\top \mathbf{v}_i) \right\|^2 - \boldsymbol{\omega}_e^\top \mathbf{K}_\omega \boldsymbol{\omega}_e - \tilde{\boldsymbol{\omega}}^\top \mathbf{C}_\omega \tilde{\boldsymbol{\omega}}.$$

where $c_R > 0$, $\mathbf{K}_\omega \succ \mathbf{0}$, $\mathbf{C}_\omega \succ \mathbf{0}$.

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(i) Standard Lyapunov theory is difficult to apply

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- (i) Standard Lyapunov theory is difficult to apply
- (ii) As $\dot{\mathcal{V}}_1$ is negative semi-definite, \mathcal{V}_1 is upper bounded,

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- (iii) Due to (ii), errors are bounded, and therefore $\ddot{\mathcal{V}}_1$ is bounded
- (iv) Due to (iii), $\dot{\mathcal{V}}_1$ is uniformly continuous in time

Case study - Barbălat

Lemma (Barbălat '59 [9])

Let $\phi : \mathbb{R}_{\geq 0} \rightarrow \mathbb{R}$ be a uniformly continuous function on its domain. If $\Phi(t) = \lim_{t \rightarrow \infty} \int_0^t \phi(\tau) d\tau$ exists and is finite, $\phi(t) \rightarrow 0$ as $t \rightarrow \infty$.



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Application of the Lemma yields asymptotic convergence to

$$\lim_{(t-t_0) \rightarrow \infty} \dot{\mathcal{V}}_1 = \mathbf{0} \Rightarrow \begin{cases} \lim_{(t-t_0) \rightarrow \infty} \omega_e = \mathbf{0} \\ \lim_{(t-t_0) \rightarrow \infty} \tilde{\omega} = \mathbf{0} \\ \lim_{(t-t_0) \rightarrow \infty} \sum_{i=1}^N k_i \mathcal{S}(\hat{\mathbf{R}}^\top \mathbf{v}_i) (\mathbf{R}_r^\top \mathbf{v}_i + \mathbf{R}^\top \mathbf{v}_i) = \mathbf{0} \end{cases}$$

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Case study - Barbălat

Lemma (Variant of Barbălat's Lemma [10, Lemma 2.2.12])

Let $f : \mathbb{R}_{\geq 0} \rightarrow \mathbb{R}$ be any differentiable function. If $f(t)$ converges to zero as $t \rightarrow \infty$ and its derivative satisfies

$$\dot{f}(t) = f_0(t) + \eta(t) \quad t \geq 0, \quad (9)$$

where $f_0 : \mathbb{R}_{\geq 0} \rightarrow \mathbb{R}$ is uniformly continuous and $\eta : \mathbb{R}_{\geq 0} \rightarrow \mathbb{R}$. If $\eta(t)$ tends to zero as $t \rightarrow \infty$, $\dot{f}(t)$ and $f_0(t)$ tend to zero as $t \rightarrow \infty$.



Case study - Barbălat

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where $f_0 : \mathbb{R}_{\geq 0} \rightarrow \mathbb{R}$ is uniformly continuous and $\eta : \mathbb{R}_{\geq 0} \rightarrow \mathbb{R}$. If $\eta(t)$ tends to zero as $t \rightarrow \infty$, $\dot{f}(t)$ and $f_0(t)$ tend to zero as $t \rightarrow \infty$.

Consider

$$\mathbf{J}\dot{\boldsymbol{\omega}}_e = f_3(t, \mathbf{R}_e, \tilde{\mathbf{R}}_e, \boldsymbol{\omega}_e, \tilde{\boldsymbol{\omega}})$$

Case study - Barbălat

Lemma (Variant of Barbălat's Lemma [10, Lemma 2.2.12])

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As $\dot{f}(t) \rightarrow 0$ and $f_0(t)$ is uniformly continuous, $f_0(t) \rightarrow 0$ as $t \rightarrow \infty$.

Case study - Barbălat

Summary from Barbălat

- First application,

$$\sum_{i=1}^N k_i \mathbf{S}(\hat{\mathbf{R}}^\top \mathbf{v}_i)(\mathbf{R}_r^\top \mathbf{v}_i + \mathbf{R}^\top \mathbf{v}_i) \triangleq f_0(t) + g_0(t) \rightarrow \mathbf{0}.$$

- Signal chasing, $f_0(t) \rightarrow 0 \Rightarrow g_0(t) \rightarrow 0$ as $(t - t_o) \rightarrow \infty$

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All solutions converge to an invariant set

$$\mathcal{S} = \left\{ (\mathbf{R}_e, \tilde{\mathbf{R}}, \boldsymbol{\omega}_e, \tilde{\boldsymbol{\omega}}) \in \text{SO}(3)^2 \times \mathbb{R}^6 \left| \begin{array}{l} \sum_{i=1}^N k_i \mathbf{S}(\mathbf{R}_r^\top \mathbf{v}_i) \hat{\mathbf{R}}^\top \mathbf{v}_i = \mathbf{0} \\ \sum_{i=1}^N k_i \mathbf{S}(\mathbf{R}^\top \mathbf{v}_i) \hat{\mathbf{R}}^\top \mathbf{v}_i = \mathbf{0} \\ \boldsymbol{\omega}_e = \mathbf{0} \\ \tilde{\boldsymbol{\omega}} = \mathbf{0} \end{array} \right. \right\}.$$

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$$\sum_{i=1}^N k_i \mathbf{S}(\hat{\mathbf{R}}^\top \mathbf{v}_i) (\mathbf{R}_r^\top \mathbf{v}_i + \mathbf{R}^\top \mathbf{v}_i) \triangleq f_0(t) + g_0(t) \rightarrow \mathbf{0}.$$

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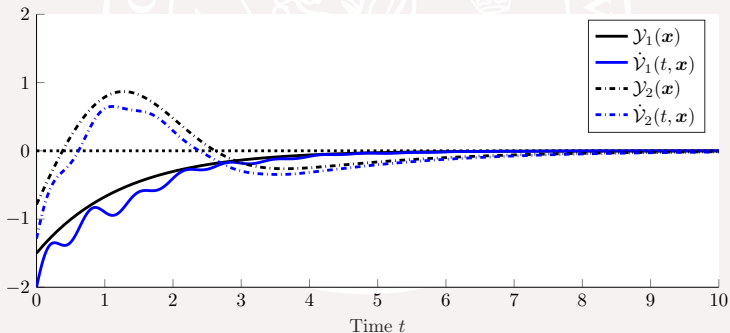
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However, convergence to \mathcal{S} is asymptotic, but not necessarily *uniform*.

Case study - Matrosov

The main idea of Matrosov

- NSD $\dot{\mathcal{V}}_1$ and non-autonomous error dynamics
- Find uniformly bounded function \mathcal{Y}_i which upper bounds $\dot{\mathcal{V}}_i$
- Satisfy nested properties on \mathcal{Y}_i
- Exact details in [3, Thm. 1 and Thm. 2].



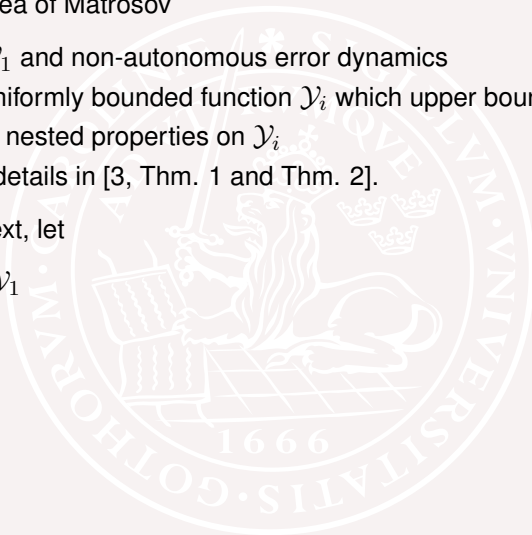
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- $\dot{\mathcal{V}}_1 \triangleq \mathcal{Y}_1$



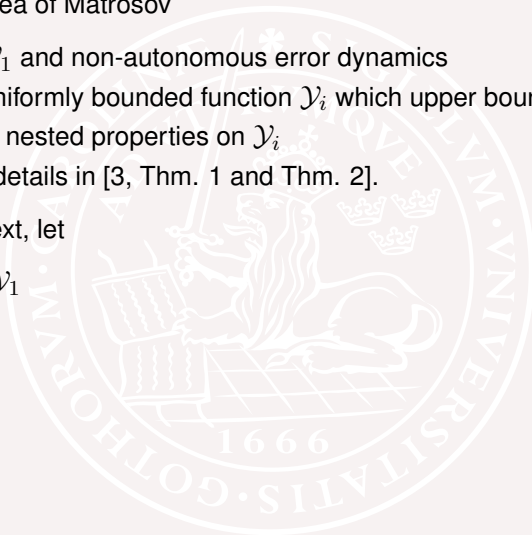
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In this context, let

- $\dot{\mathcal{V}}_1 \triangleq \mathcal{Y}_1$
- $\mathcal{V}_2 = \omega_e^\top \sum_{i=1}^N k_i \mathbf{S}(\mathbf{R}_r^\top \mathbf{v}_i) \hat{\mathbf{R}}^\top \mathbf{v}_i.$

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- Then, plugging in the error dynamics,

$$\frac{d\mathcal{V}_2}{dt} \leq - \left\| \sum_{i=1}^N k_i \mathbf{S}(\mathbf{R}_r^\top \mathbf{v}_i) \hat{\mathbf{R}}^\top \mathbf{v}_i \right\|^2 + M_3 \left\| \begin{bmatrix} \omega_e \\ \tilde{\omega} \end{bmatrix} \right\| + M_4 \left\| \begin{bmatrix} \omega_e \\ \tilde{\omega} \end{bmatrix} \right\|^2 \triangleq \mathcal{Y}_2,$$

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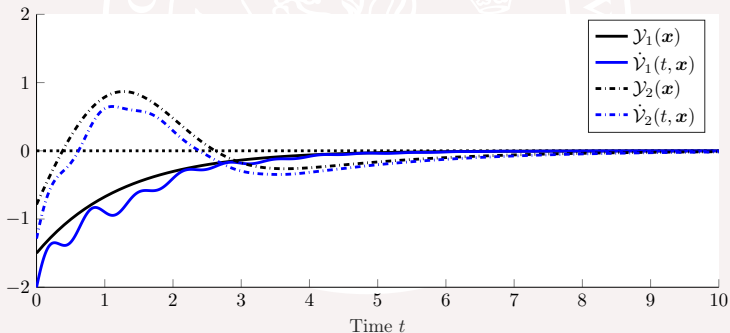
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- $\mathcal{Y}_1 = 0 \Rightarrow \omega_e = \tilde{\omega} = \mathbf{0} \Rightarrow \mathcal{Y}_2 \leq 0$, and
- $\mathcal{Y}_1 = \mathcal{Y}_2 = 0 \Rightarrow (\mathbf{R}_e, \tilde{\mathbf{R}}, \omega_e, \tilde{\omega}) \rightarrow \mathcal{S}.$

Case study - Matrosov

The main idea of Matrosov

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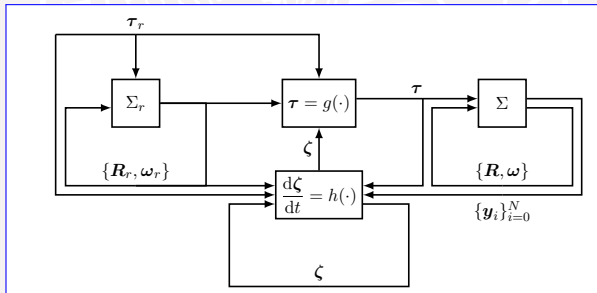
Case study - Summary

Barbalat + Signal chasing + Matrosov

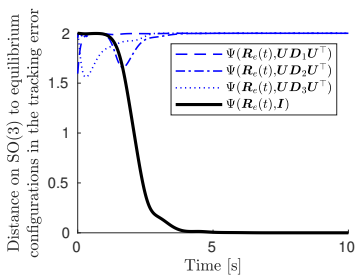
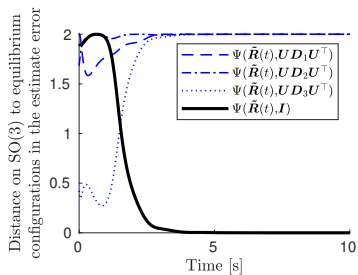
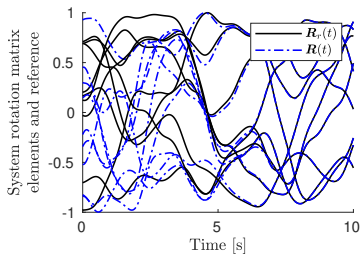
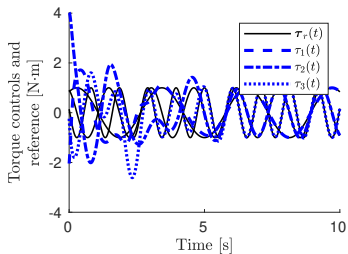
- Uniform asymptotic convergence of $(\mathbf{R}_e, \tilde{\mathbf{R}}, \omega_e, \tilde{\omega}) \rightarrow \mathcal{S}$

It is also possible to show

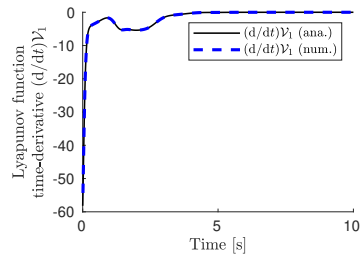
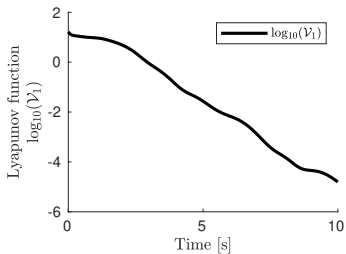
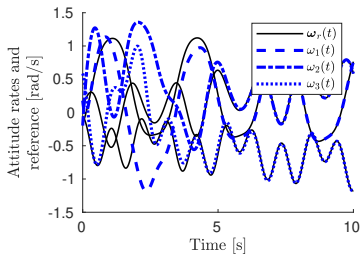
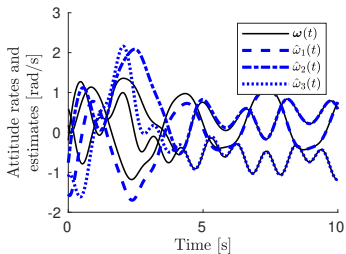
- By conditions on $\{(k_i, \mathbf{v}_i)\}_{i=1}^N$: \mathcal{S} contains 13 isolated equilibrium points
- By local linearization : $(\mathbf{I}, \mathbf{I}, \mathbf{0}, \mathbf{0}) \in \mathcal{S}$ is UAGAS
- By local linearization : $(\mathbf{I}, \mathbf{I}, \mathbf{0}, \mathbf{0}) \in \mathcal{S}$ is ULES



Case study - Simulation



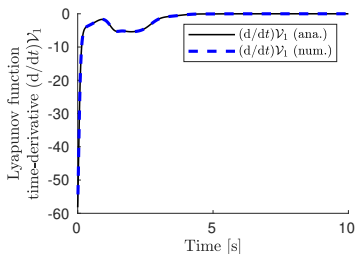
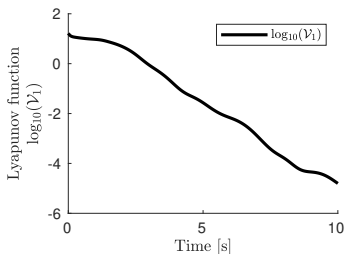
Case study - Simulation



Case study - Simulation

Recall, the proposed Lyapunov function

$$\mathcal{V}_1 = \sum_{i=1}^N \frac{k_i}{2} \|\mathbf{R}_e \tilde{\mathbf{R}}^\top \mathbf{v}_i - \mathbf{v}_i\|^2 + \frac{1}{2} \boldsymbol{\omega}_e^\top \mathbf{J} \boldsymbol{\omega}_e + \sum_{i=1}^N \frac{k_i}{2} \|\tilde{\mathbf{R}} \mathbf{v}_i - \mathbf{v}_i\|^2 + \frac{1}{2} \tilde{\boldsymbol{\omega}}^\top \mathbf{J} \tilde{\boldsymbol{\omega}}.$$

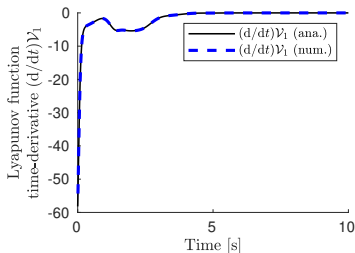
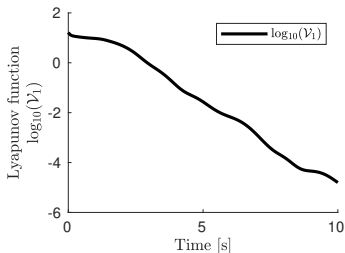


Case study - Simulation

- The Lyapunov function time derivative in the errors (black)

$$\dot{\mathcal{V}}_1 = -c_R \left\| \sum_{i=1}^N k_i \mathcal{S}(\hat{\mathbf{R}}^\top \mathbf{v}_i) (\mathbf{R}_r^\top \mathbf{v}_i + \mathbf{R}^\top \mathbf{v}_i) \right\|^2 - \boldsymbol{\omega}_e^\top \mathbf{K}_\omega \boldsymbol{\omega}_e - \tilde{\boldsymbol{\omega}}^\top \mathbf{C}_\omega \tilde{\boldsymbol{\omega}}.$$

- And evaluated from \mathcal{V}_1 by numerical differentiation (blue)



Conclusions

Summary

- Separation principle and peaking
- Uniform stability and robustness
- Tools from Lyapunov, Barbalat, and Matrosov
- Making sense of 27 error signals
- Code: `AerialVehicleControl.jl` [11]
- More: ACC Wed, 10.15 and 11.00 (UTC -5)

Thank you for listening!

[11] M. Greiff, *AerialVehicleControl.jl, nonlinear and robust UAV control system synthesis*, github.com/mgreiff/AerialVehicleControl.jl, 2020


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